

Fracture characteristics of Fly Ash reinforced A356 alloy composites

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Abstract: Use of Aluminium alloys as a substitution for auto parts made of ferrous alloys, has many positive aspects such as reduction of mass, lower fuel consumption and therefore reduced pollution. Several authors proved that addition a reinforcing phase to these alloys can significantly improve the tribological properties. The fly ash that comes out as an industrial solid waste can be utilized as reinforcing agent instead of dumping. In the present work A356 alloy reinforced with fly ash in 5%, 10% and 15% by weight are studied for their mechanical properties, machinability and analyzed metallographically for fracture and failure characteristics.

Key words: Fly ash reinforced MMCs; cutting forces; machining properties; mechanical properties; fracture toughness; fractography.

1. INTRODUCTION

Composite materials are formed by mixing of two or more materials or phases of the same material. The composite has completely new, different and better characteristics compared to its constituents. The components do not blend with each other nor do they get dissolved, so there is a visible difference between them. Constituent materials have significantly different physical and chemical properties that remain separate and distinct within the finished structure. Essentially, the composites consist of the base (the matrix), which forms the continuous phase and whose content is much higher in comparison to that of the other materials, and reinforcement, i.e. material with which the desired properties of composites are to be achieved. The reinforcement forms the discontinuous phase. While the reinforcing material usually carries the major amount of load, the matrix enables the load transfer by holding them together.

Aluminium has useful properties such as high strength, ductility, high thermal and electrical conductivity but has low stiffness. Fly ash, on the other hand, is stiffer and stronger and has excellent high temperature resistance but is brittle in nature.

Fly ash usually refers to ash produced during combustion of coal and is captured by electrostatic precipitators. It is one of the most inexpensive and low density reinforcements available in large quantities as solid waste by-product during combustion of coal in thermal power plants.

In the present study, A 356 with the theoretic density of 2760 kg/m³ was used as a matrix material. Fly ash particulates with an average size of 25µm are used as reinforcement material. Magnesium was selected as a wetting agent to improve wettability between the matrix and the reinforcements during production of the composites.

II. EXPERIMENTAL METHODS

Fly Ash is added as reinforcement in proportions (5%, 10% and 15% by weight) to A356 alloy. The process used for production is stir casting technique.

2.0 Composite fabrication: A356 Aluminium alloy charge was melted in graphite crucible at 700 ± 20°C stir casting

machine (Figure 1). The reinforcement fly ash was preheated to remove moisture at 800°C for 1 h before it was incorporated into the melt. A degassing tablet (coverall powder) was added to reduce the porosity. To enhance the wettability between the matrix and the reinforcements 1% magnesium by weight was added to the melt. The molten metal was stirred at the speed of 700 rpm with multi bladed stirrer made up of stainless steel coated with ceramic. The preheated fly ash particles were introduced into the molten metal at a constant rate. The stirring was continued for 10 more minutes after the completion of particle feeding. The melt mixture was allowed to maintain at 700°C after this stage for 10 minutes without stirring and then was poured into the mould preheated at 500°C for 30 min to obtain uniform solidification. T6 treatment was given to the cast test specimens following homogenization at 200°C for 20 hrs. The cycle of T6 heat treatment process is carried out by solutionizing for 8 h at (535±5)⁰C and then quenched in water at ambient temperature and finally artificially aged at 180 °C for 6 h followed by air cooling.



Figure 1: Stir casting machine

2.1 Tensile Tests: Tensile tests were conducted at BDL laboratory in accordance to ASTM E8 standard specifications on Ø 12.5 mm cylindrical specimens. A 400 kN Instron make UTM (Figure 2) was used for experimentation. The ram speed 10 mm/min was maintained during the test.



Figure 2: 400 kN Instron make UTM

2.2 Cutting Force Measurements: Kistler’s make tool dynamometer (Figure 3) was used for measurement of cutting forces were measured on facility at GITAM University.



Figure 3: Tool Dynamometer set-up

2.3 Fracture Toughness Tests: The newly commissioned 400 kN Instron Make fatigue testing machine (Figure 4) at GITAM University was used for conducting fracture toughness tests on SENB specimens according to ASTM E399.



Figure 4: Fracture toughness experiment on 100 kN Instron

Fatigue Testing Machine with SENB specimen

2.4 Fractography: Fractographic studies were carried out on Hitachi make SEM in central facility of Osmania University.

III. RESULTS AND DISCUSSION

A considerable amount work has been reported on the properties and characterization of fly ash reinforced A356 alloy literature. But the evidence of fracture toughness are not found. In the present work an attempt is made to study and characterize these properties.

3.1 Mechanical properties: The yield and ultimate strengths of the composite were plotted against fly ash reinforcement in figure 5. A slight increase in the ultimate strength was observed initially till 10% fly ash addition and a slight decrease beyond. The yield strength increased continuously with particle reinforcement. The Young’s showed pattern similar to yield strength (figure 6) as it is the measure of elastic deformation of tension test. In contrast to yield strength, the percentage elongation decreased with fly ash (figure 7).

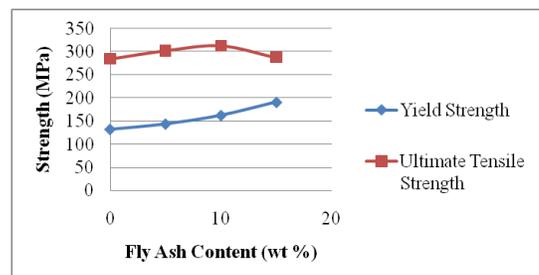


Figure 5: Variation of Strength with Fly Ash content

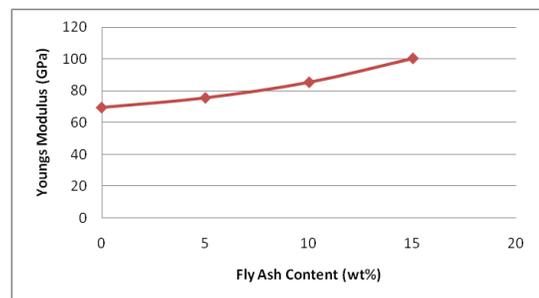


Figure 6: Variation of Young’s Modulus with Fly Ash content

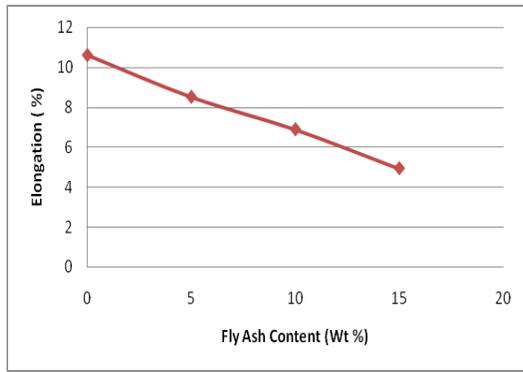


Figure 7: Variation of Elongation with Fly Ash content

3.2 Machining Properties: Three components of machining forces during oblique turning process using HSS tool with 20-15-12-10-5-5-1 signature. The approach angle selected is 70° . The parameters used in cutting were cutting speed = 24.2 m/min (rpm = 350) & 38.72 m/min (rpm = 560), Feed = 0.14 & 0.16 mm/rev, depth of cut = 2 mm, diameter = 22 mm and length of turning = 140 mm respectively. The figures 8 through 10 depict the variation of maximum thrust, feed and radial force with wt% of fly ash reinforced. The maximum power consumption Vs. fly ash content is plotted in figure 11. Except few abnormalities, in most of the cases these cutting forces and power requirements reduced with increase in fly ash content to 10% and increased beyond this extent.

Figure 8: Variation of Maximum Thrust Force with Fly Ash content

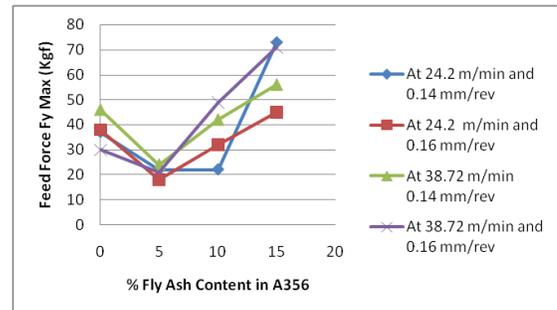


Figure 9: Variation of Maximum Feed Force with Fly Ash content

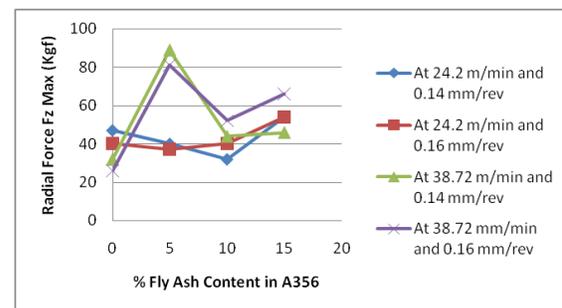


Figure 10: Variation of Maximum Radial force with Fly Ash content

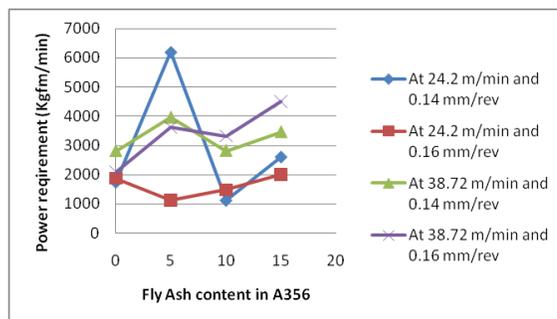
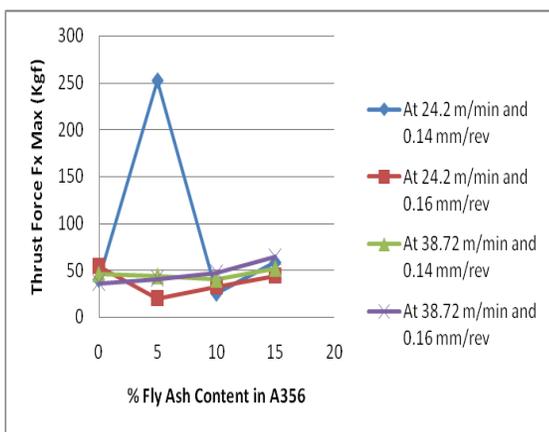


Figure 11: Variation of Maximum power requirement with Fly Ash content

3.3 Plain Strain Fracture Toughness: The K_{IC} tests were conducted using 100 kN Instron make fatigue testing machine on single edge notched bend (SENB) specimen as shown in figure 12. The dimensions were selected as permitted by ASTM E399 and the availability of casting dies. The specimen dimensions are $B = 8$ mm, $W = 32$ mm, Span = 70 mm and notch depth = 1.5 mm. The pre-cracking was limited to $a = 2$ mm.

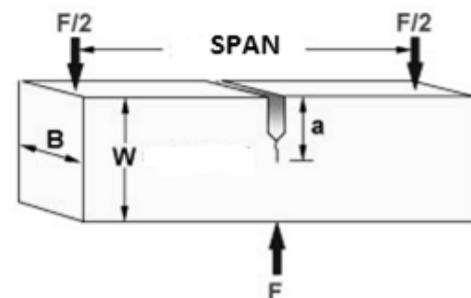


Figure 12: Single Edge Notched Bend (SENB) specimen

The variation of plain strain fracture toughness (K_{IC}) values for composites Vs. fly ash content were plotted in figure 13. A considerable drop was observed with addition of fly ash reinforcement in the beginning. Further increase in our experimental range had no effect.

3.4 Fractography: The SEM studies of fractographs taken in various locations for these composites after fracturing. The analysis is tabulated in table 1 through 5. The first location termed as zone-I is the zone of pre-cracking; the second location is termed as zone-II is the immediate neighborhood of pre-crack in the direction of crack propagation; the third zone is transition zone and indicated as zone-III; next to it is zone-IV, the transgranular cleavage region followed by zone-V, the intra granular cleavage region. The magnification used was $X10^5$.

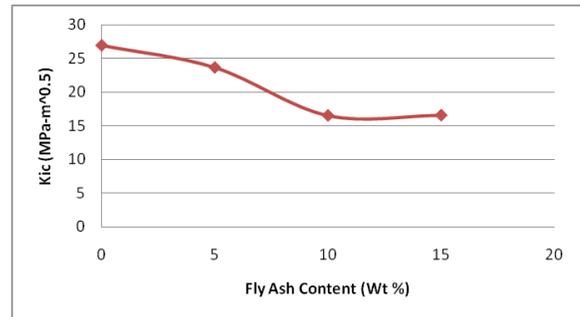


Figure 13: Variation of Plain Strain Fracture toughness with Fly Ash content

Table 1: SEM fractographs of Zone-I

Composite	Image	Prediction
Monolithic A356		Fatigue intrusions and protrusions associated with pre-cracking.
A356 + 5% Fly Ash		Fatigue intrusions and protrusions associated with pre-cracking. The fly ash particles are seen.

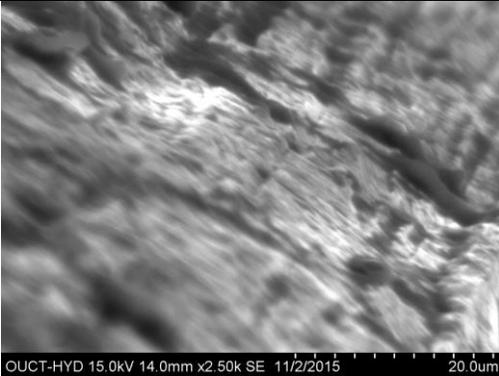
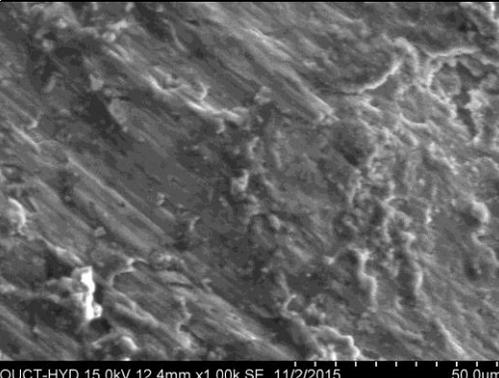
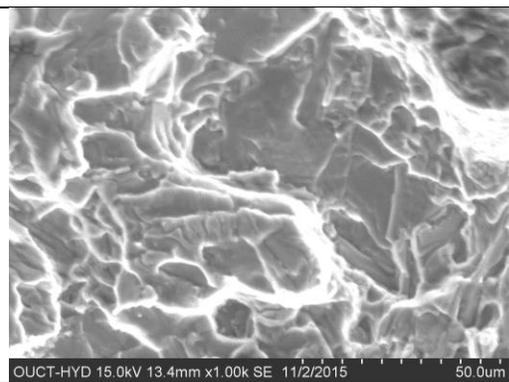
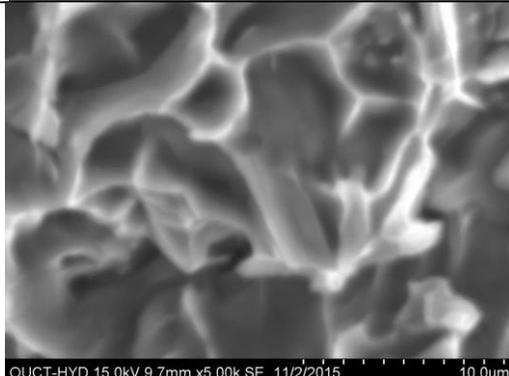
<p>A356 + 10 % Fly Ash</p>		<p>Appearance of dimple formation in fatigue pre-cracking region.</p>
<p>A356 + 15% Fly Ash</p>		<p>Fly ash particles embedded in matrix with fatigue micro striations</p>

Table 2: SEM fractographs of Zone-II

Composite	Image	Prediction
<p>Monolithic A356</p>		<p>Stable crack growth appears surrounding the micro pores.</p>
<p>A356 + 5% Fly Ash</p>		<p>Stable crack growth in the vicinity of reinforcement interface.</p>

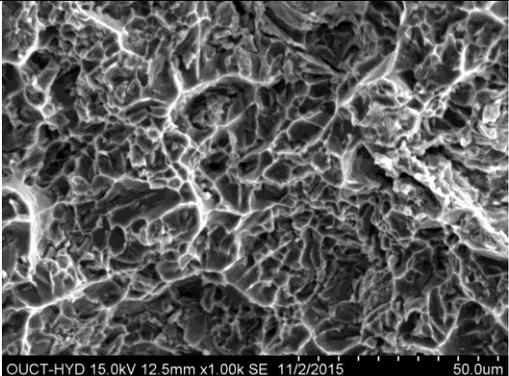
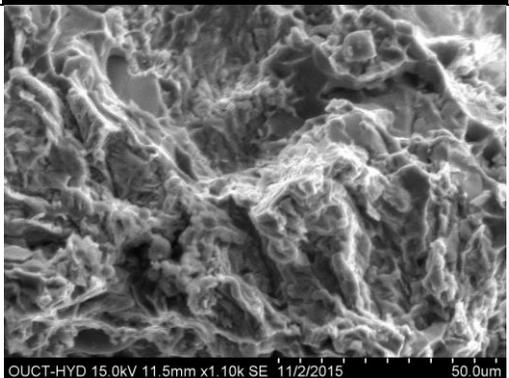
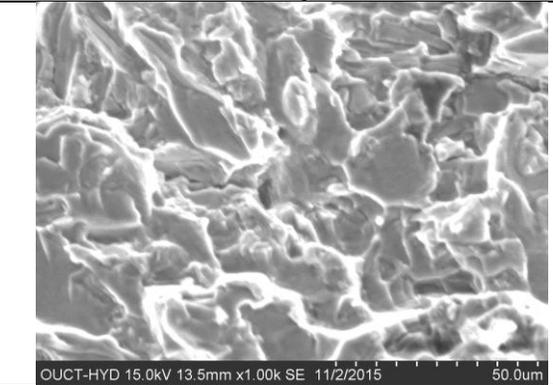
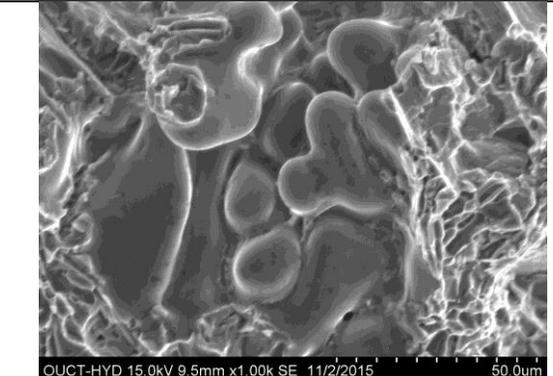
<p>A356 + 10 % Fly Ash</p>	 <p>OUCT-HYD 15.0kV 12.5mm x1.00k SE 11/2/2015 50.0um</p>	<p>Stable crack growth due to many interfaces at higher fly ash content.</p>
<p>A356 + 15% Fly Ash</p>	 <p>OUCT-HYD 15.0kV 11.5mm x1.10k SE 11/2/2015 50.0um</p>	<p>Stable crack growth at reinforcement interface. Large size micro cracks are associated with particle agglomeration due to very high reinforcement.</p>

Table 3: SEM fractographs captured in Zone-III

Composite	Image	Prediction
<p>Monolithic A356</p>	 <p>OUCT-HYD 15.0kV 13.5mm x1.00k SE 11/2/2015 50.0um</p>	<p>In this zone the failure mode is in transition from ductile mode to brittle mode.</p>
<p>A356 + 5% Fly Ash</p>	 <p>OUCT-HYD 15.0kV 9.5mm x1.00k SE 11/2/2015 50.0um</p>	<p>In this zone the failure mode is in transition from ductile mode to brittle mode.</p>

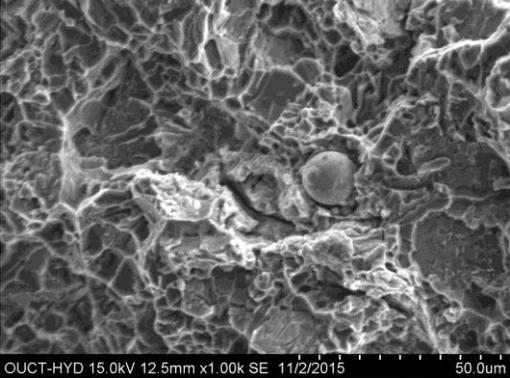
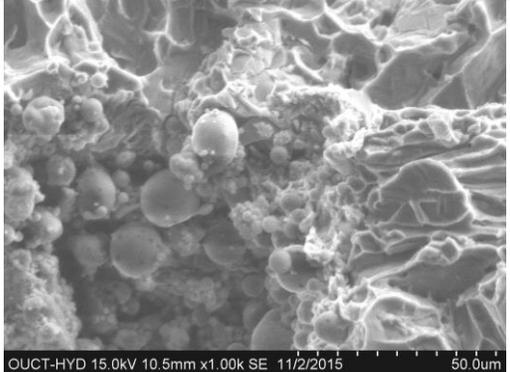
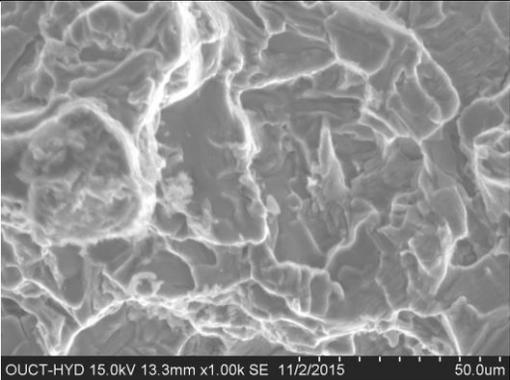
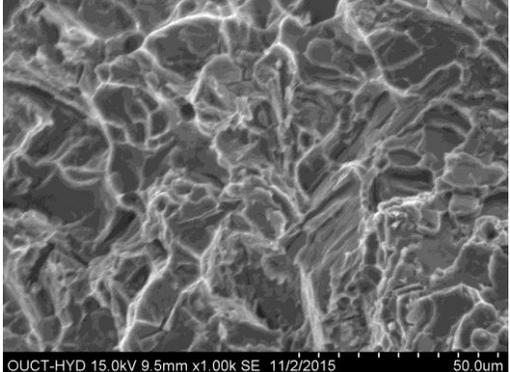
<p>A356 + 10 % Fly Ash</p>		<p>In this zone the failure mode is in transition from ductile mode to brittle mode.</p>
<p>A356 + 15% SiC</p>		<p>In this zone the failure mode is in transition from ductile mode to brittle mode.</p>

Table 4: SEM fractographs captured in Zone-IV

Composite	Image	Prediction
<p>Monolithic A356</p>		<p>This Zone is visualized by transgranular cleavage.</p>
<p>A356 + 5% Fly Ash</p>		<p>This Zone is visualized by transgranular cleavage.</p>

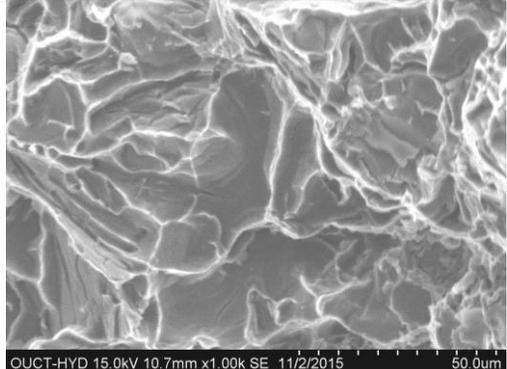
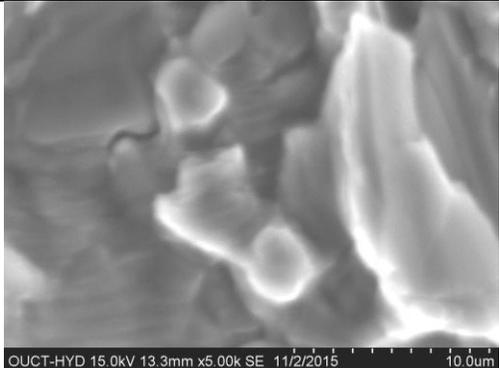
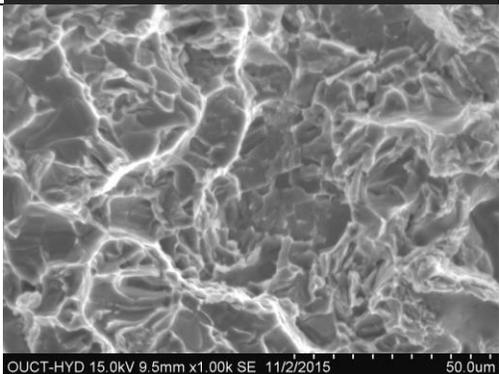
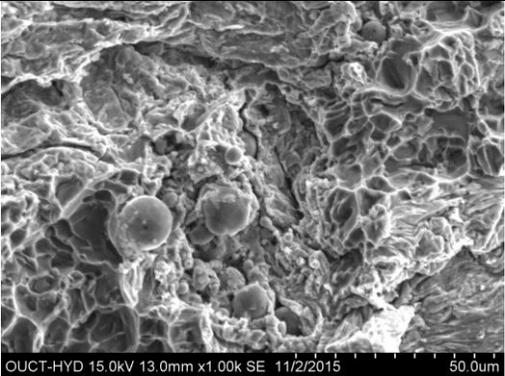
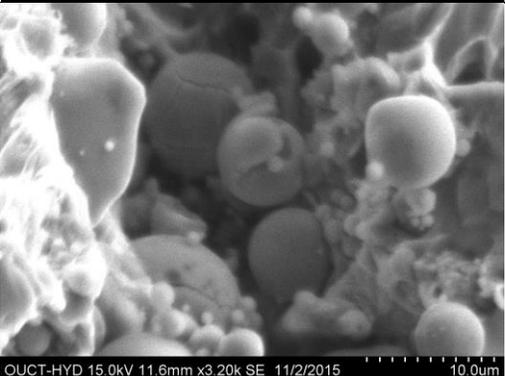
<p>A356 + 10 % Fly Ash</p>	 <p>Micrograph showing a large, rounded, smooth particle (likely a fly ash particle) surrounded by a fractured matrix. The particle is spherical and has a smooth surface. The matrix is highly fractured, showing transgranular cleavage. The scale bar indicates 10.0um.</p>	<p>This Zone is visualized by transgranular cleavage.</p>
<p>A356 + 15% Fly Ash</p>	 <p>Micrograph showing a highly fractured, rough surface with a network of cracks. The surface is highly irregular and shows transgranular cleavage. The scale bar indicates 50.0um.</p>	<p>This Zone is visualized by transgranular cleavage.</p>

Table 5: SEM fractographs captured in Zone-V

Composite	Image	Prediction
<p>Monolithic A356</p>	 <p>Micrograph showing a highly fractured, rough surface with a network of cracks. The surface is highly irregular and shows transgranular cleavage. The scale bar indicates 10.0um.</p>	<p>The highly fractured rough surface in brittle zone of inter-granular.</p>
<p>A356 + 5% Fly Ash</p>	 <p>Micrograph showing a highly fractured, rough surface with a network of cracks. The surface is highly irregular and shows transgranular cleavage. The scale bar indicates 50.0um.</p>	<p>The highly fractured rough surface in brittle zone of inter-granular.</p>

<p>A356 + 10 % Fly Ash</p>		<p>The highly fractured rough surface in brittle zone of inter-granular.</p>
<p>A356 + 15% Fly Ash</p>		<p>The highly fractured rough surface in brittle zone of inter-granular.</p>

IV.CONCLUSIONS

The machining and fracture characteristics of fly ash reinforced A356 alloy samples with varying reinforcement content were studied after successfully fabricating. Addition 5 to 10% of fly ash by weight as reinforcement had lead to better machining properties. Further advantage was not gained with higher addition. The mechanical properties such as yield stress, ultimate stress too had similar effect. The major loss in ductility and K_{IC} values has occurred with addition of fly ash between 5 to 10% by weight. Further addition had not worsened these properties. The best machinability and optimum mechanical properties were achieved with addition of 5 to 10% fly ash reinforcement to A356 alloy.

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